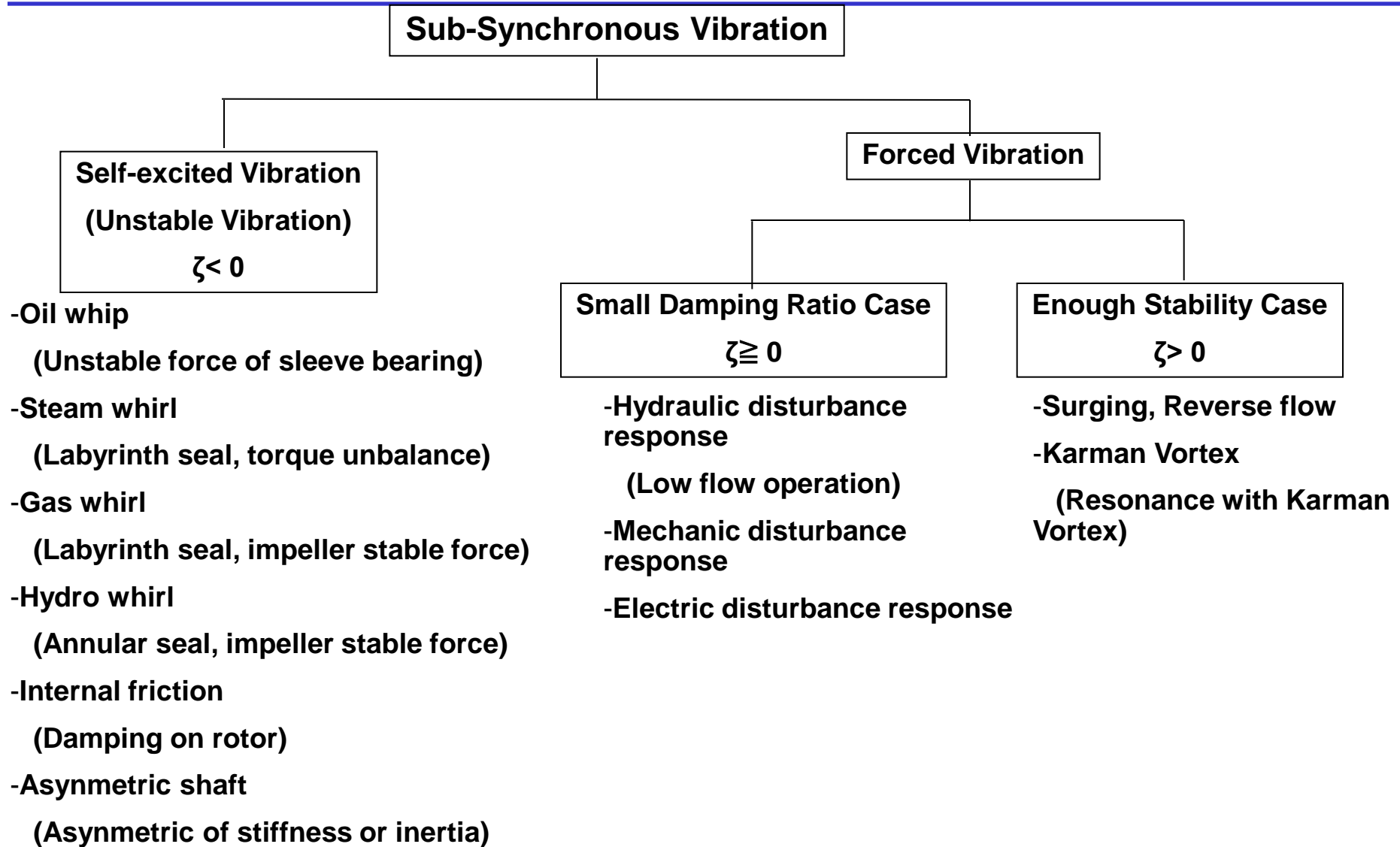


***40th Turbomachinery Symposium
Abstract***

***Coupled Sub-Synchronous Vibration
of Lateral and Axial Directions***

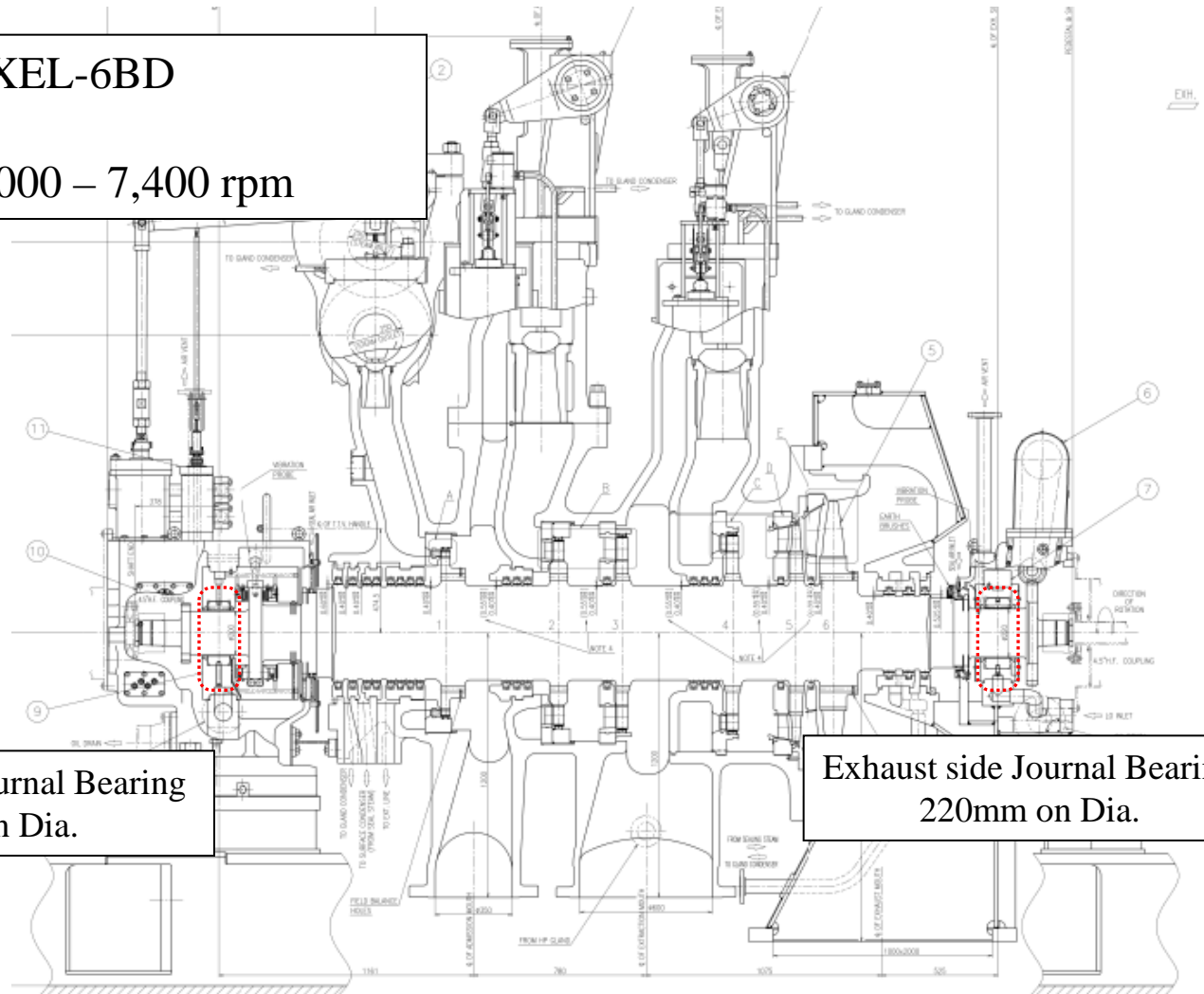
***Mitsubishi Heavy Industries Compressor Corporation
Daisuke Kiuchi***

Classification of Sub-Synchronous Vibration



539

Operating Range: 5,000 – 7,400 rpm



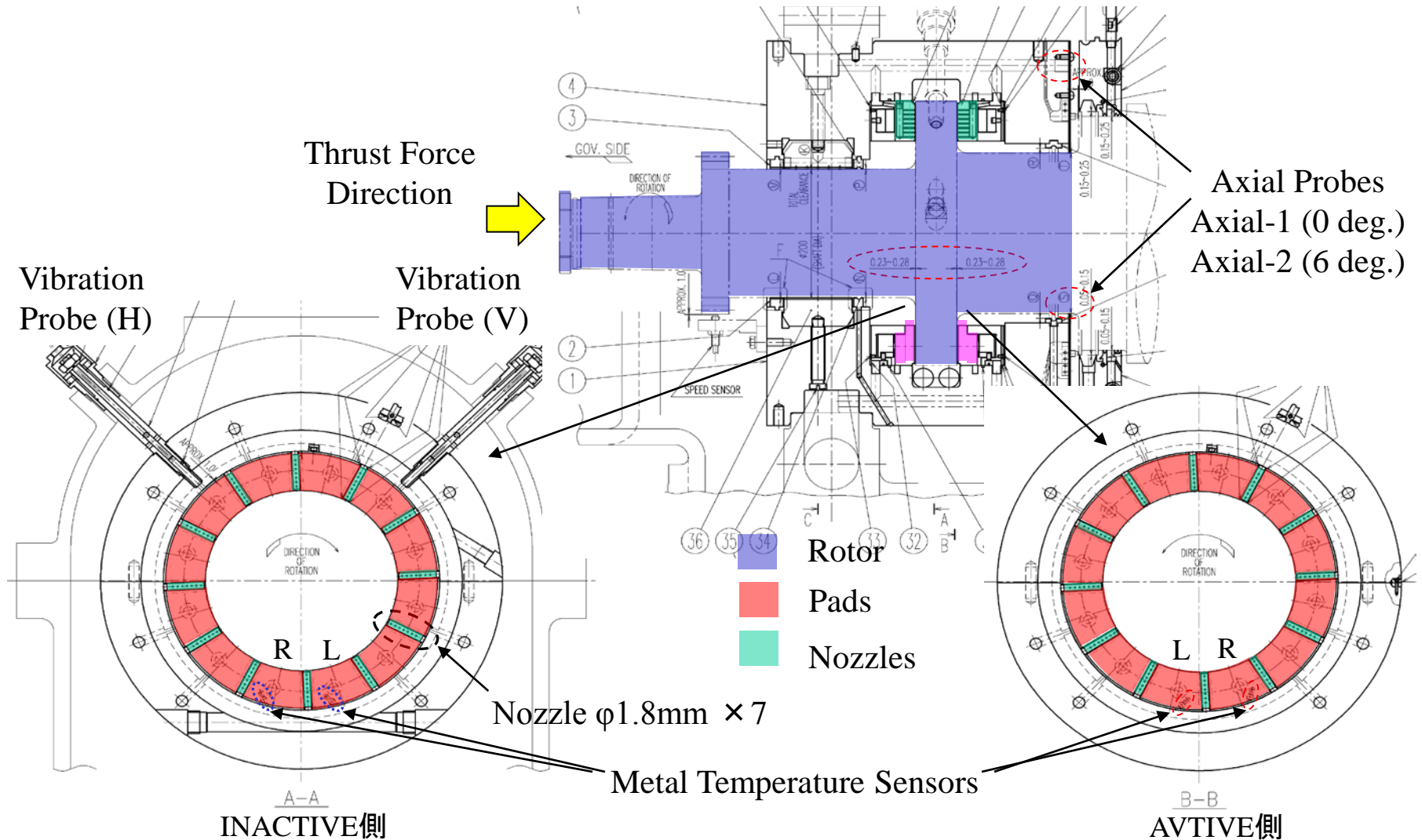
Governor side Journal Bearing
200mm on Dia.

Exhaust side Journal Bearing
220mm on Dia.

Coupled Sub-Synchronous Vibration Lateral and Axial Directions

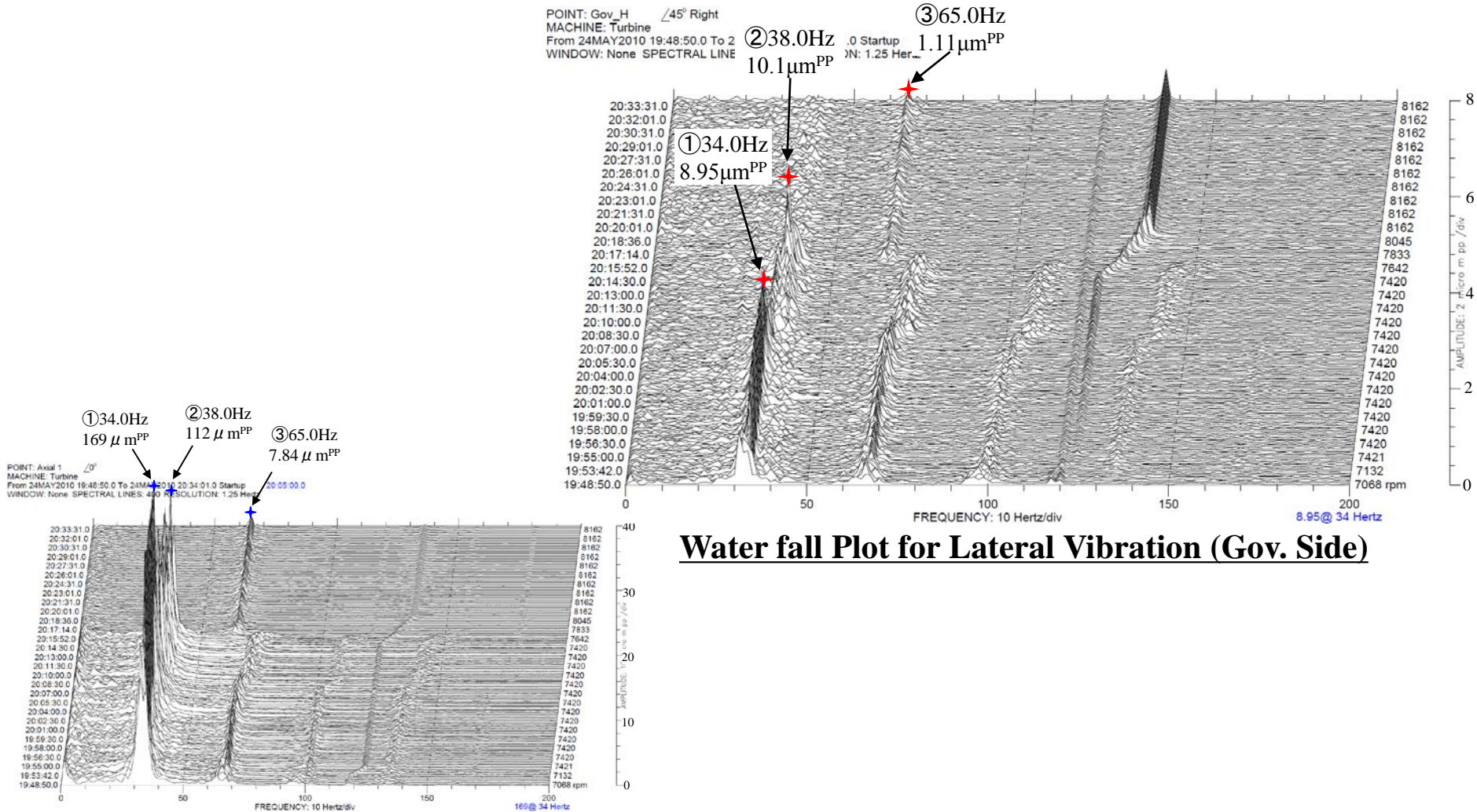
Case-1

Configuration of Bearings on Governor Side



Coupled Sub-Synchronous Vibration Lateral and Axial Directions

Case-1



Water fall Plot for Axial Vibration

The same sub-synchronous frequency was observed for lateral and axial direction.

Coupled Sub-Synchronous Vibration Lateral and Axial Directions

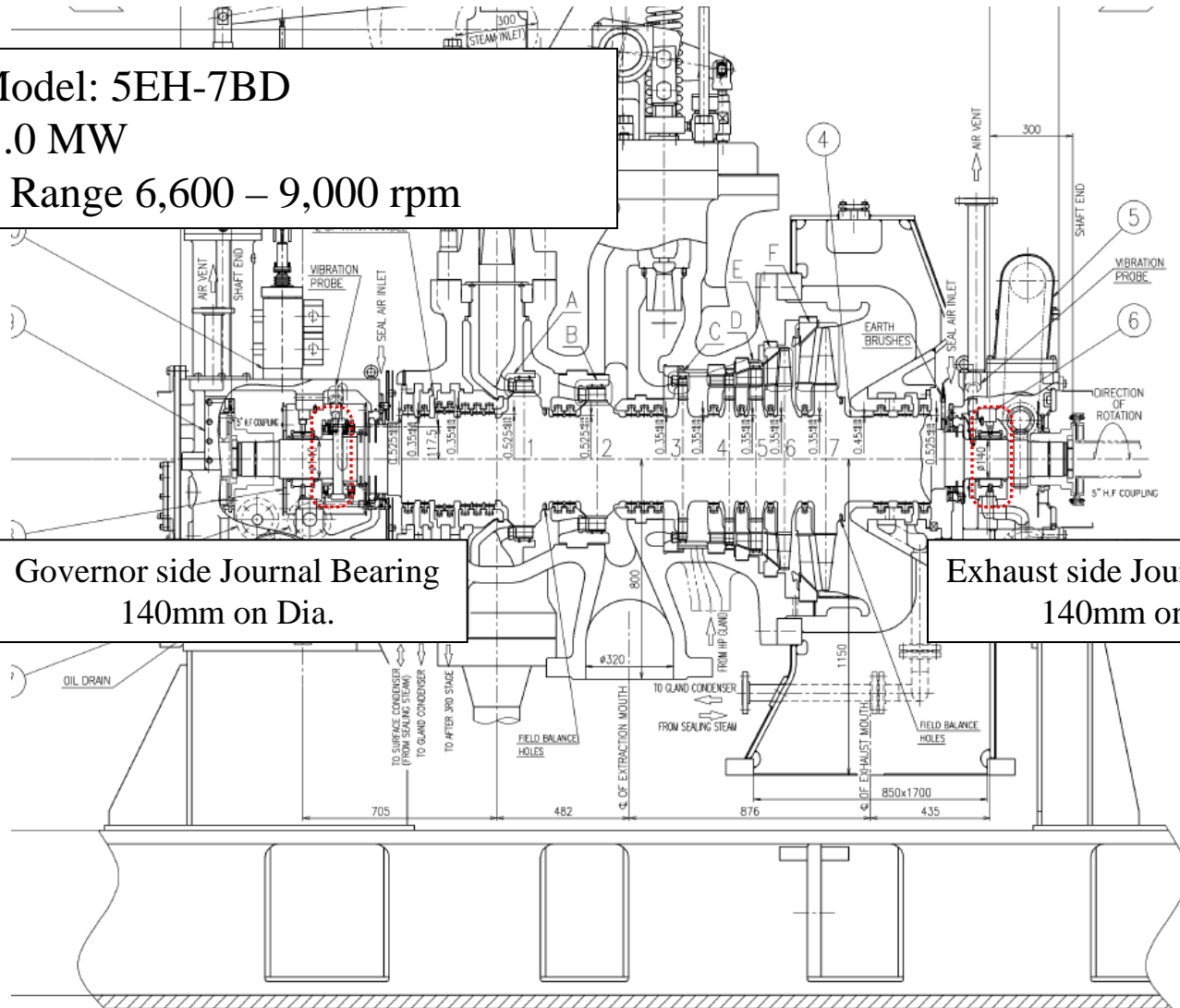
Case-2

Another sub-synchronous vibration was observed during the shop mechanical running test of the following steam turbine.

Turbine Model: 5EH-7BD

Power: 43.0 MW

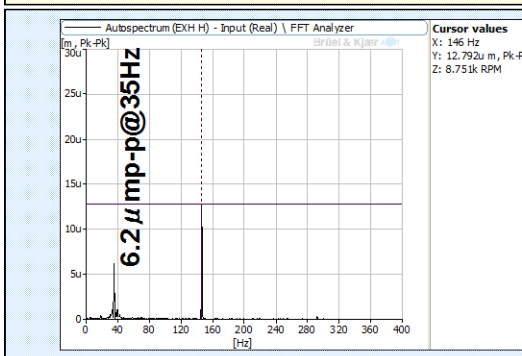
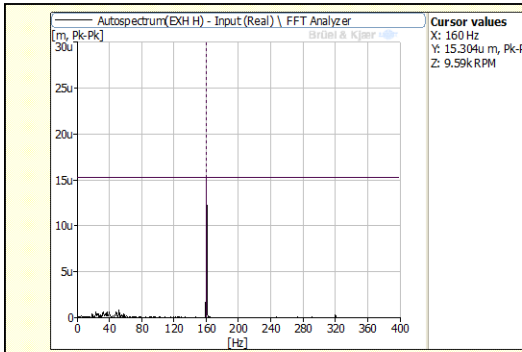
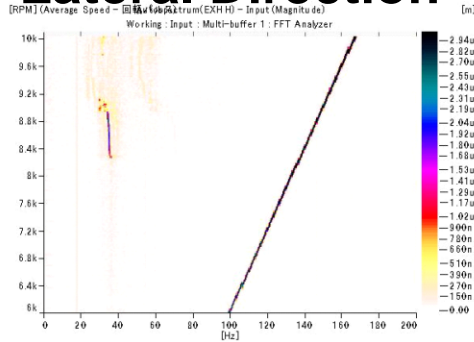
Operating Range 6,600 – 9,000 rpm



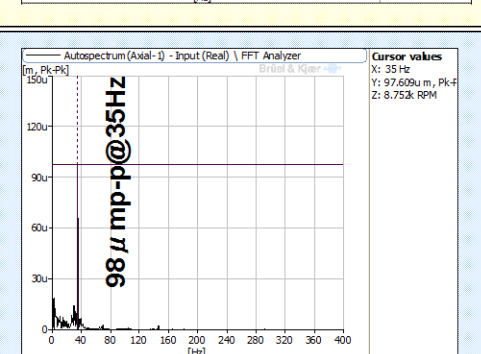
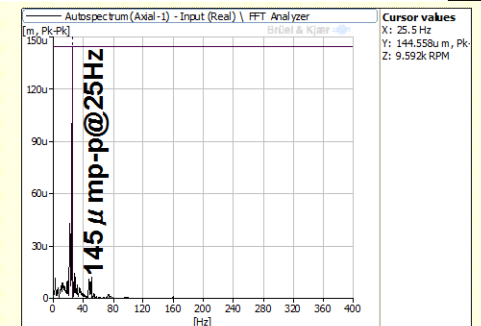
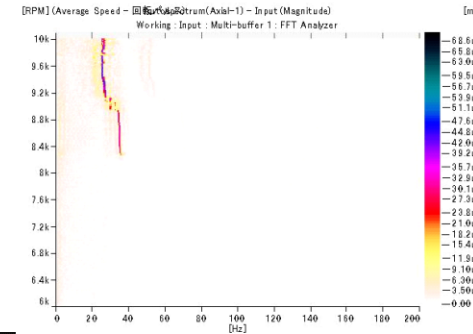
Coupled Sub-Synchronous Vibration Lateral and Axial Directions

Case-2

Lateral Direction



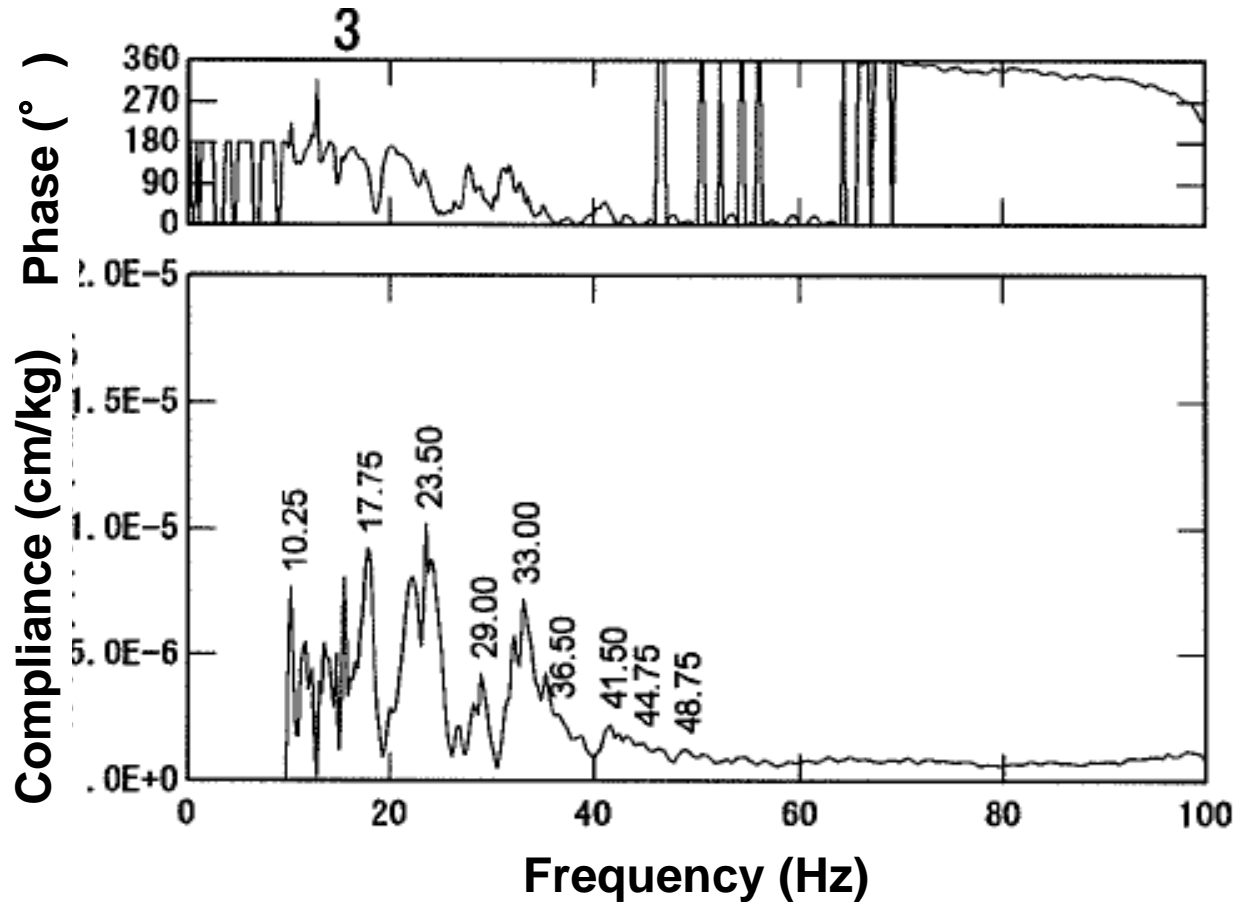
Axial Direction



The same sub-synchronous frequency was observed for lateral and axial direction.

Case-2

Hammering Test for Pedestal

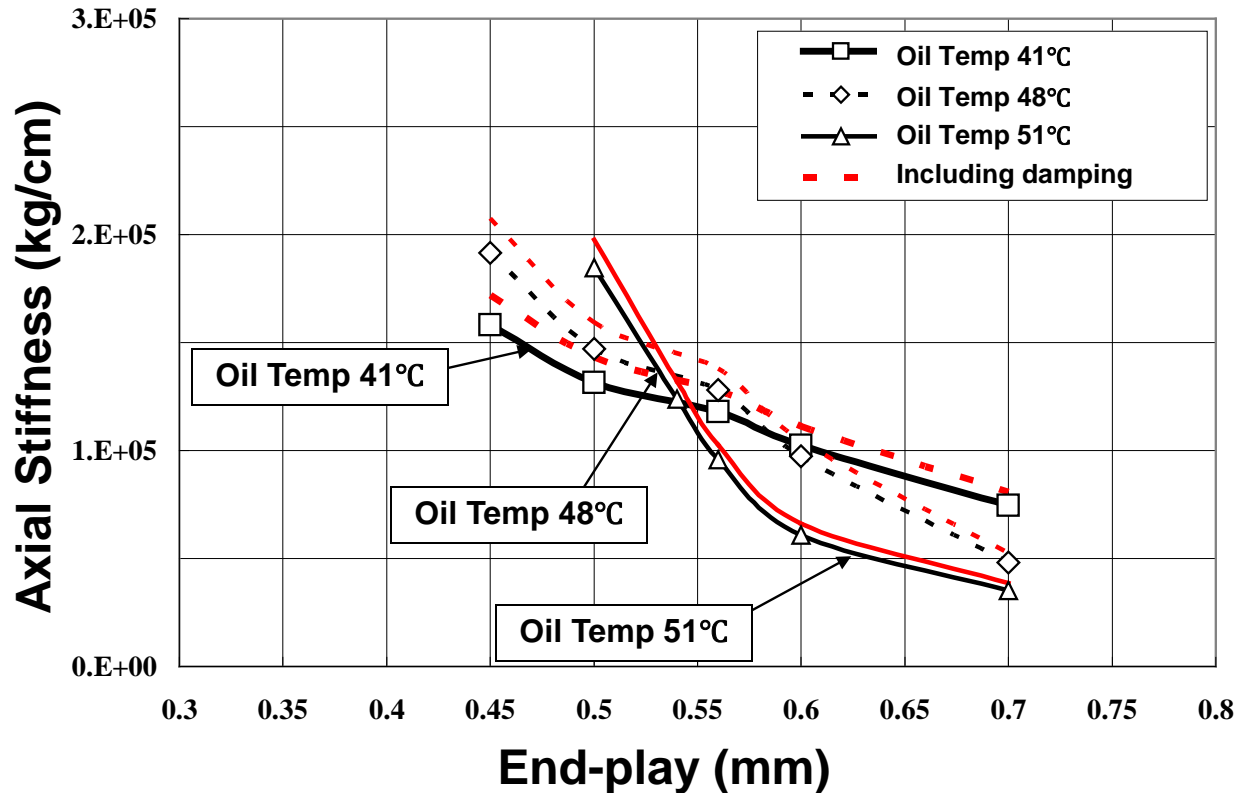


Natural Frequency of 35 - 40Hz could not be found.

Coupled Sub-Synchronous Vibration Lateral and Axial Directions

Case-2

Simulation Results of Thrust Bearing Stiffness



Axial Natural Frequency

$$m \approx 2,000 \text{ kg}$$

$$k \approx 1 \times 10^5 \text{ kgf / cm}$$

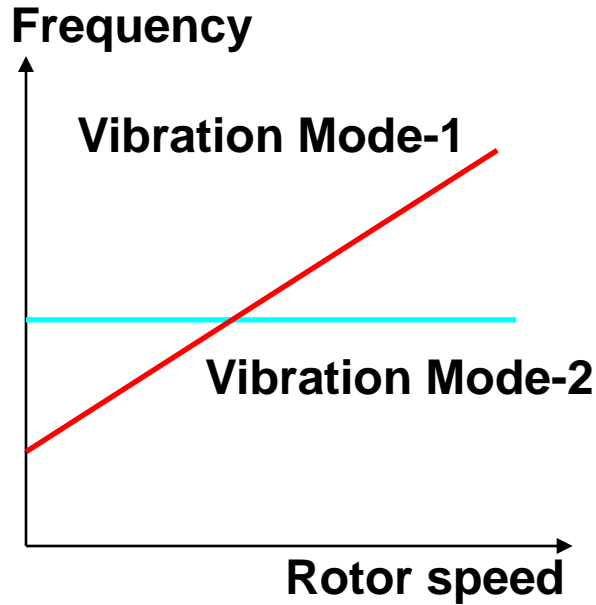
$$f = \frac{1}{2\pi} \sqrt{\frac{k}{m}}$$

$$= \frac{1}{2\pi} \sqrt{\frac{1 \times 10^5 \times 980}{2,000}}$$

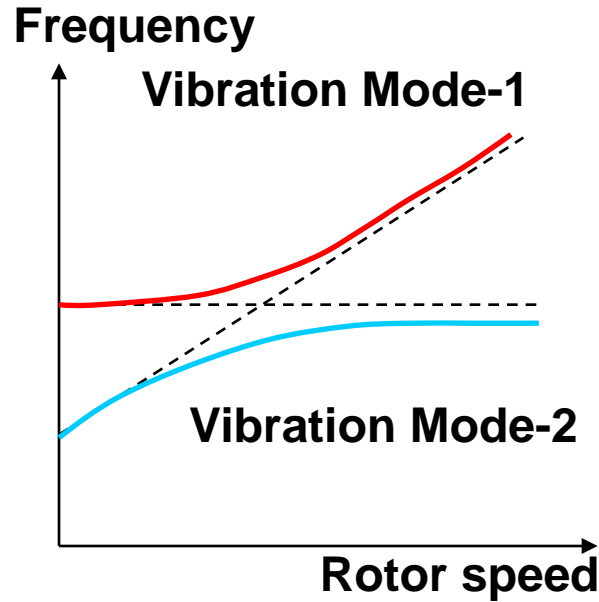
$$= 35.2 \text{ Hz}$$

Agreed with sub-synchronous vibration frequency.

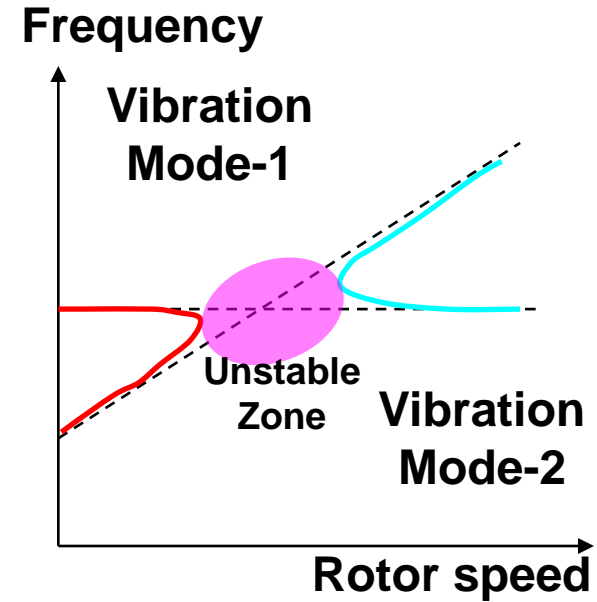
Unstable Coupled Vibration of Lateral and Axial Directions



**No Energy
Interaction**

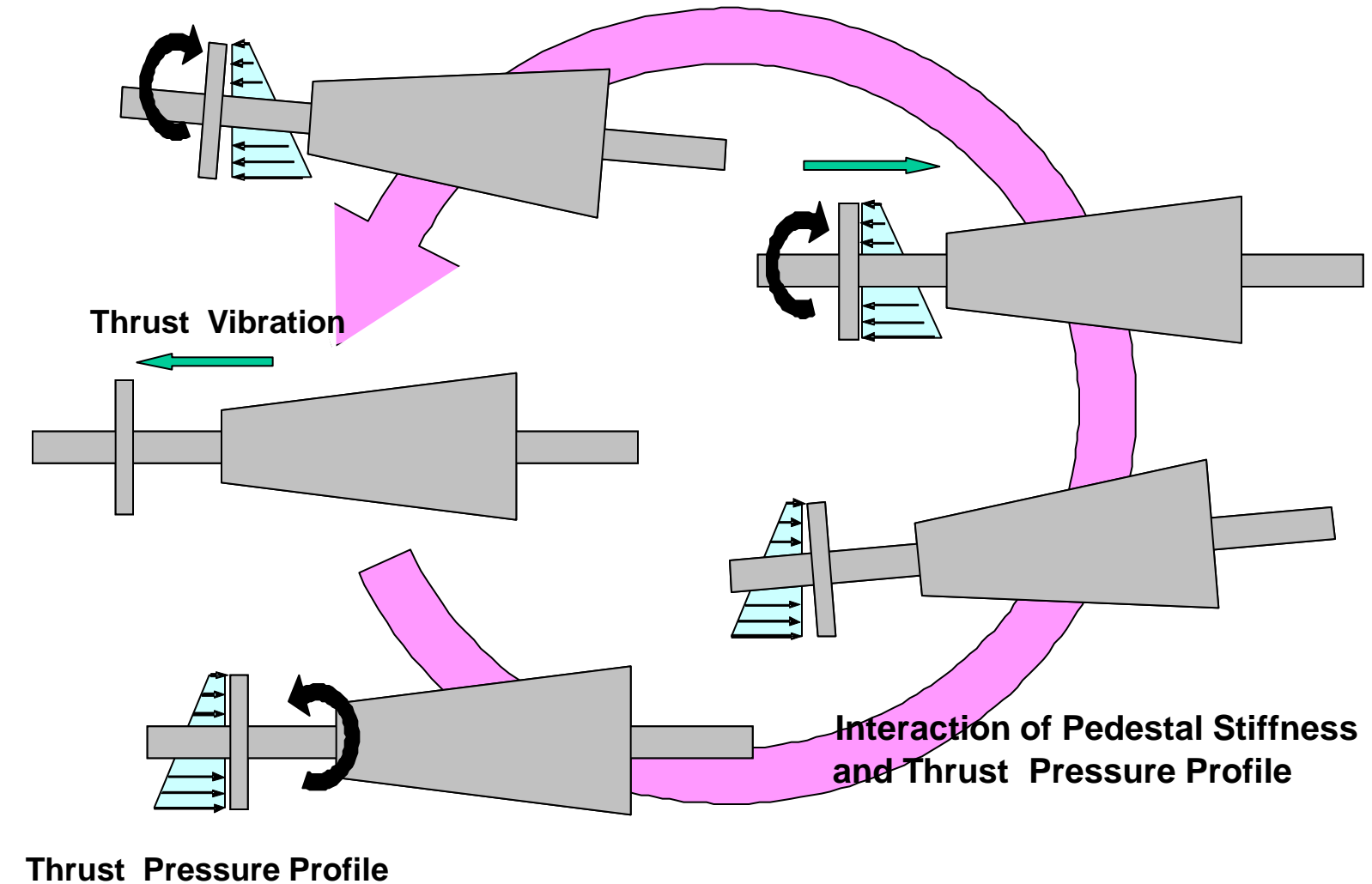


**General
Stable
Coupled Vibration**



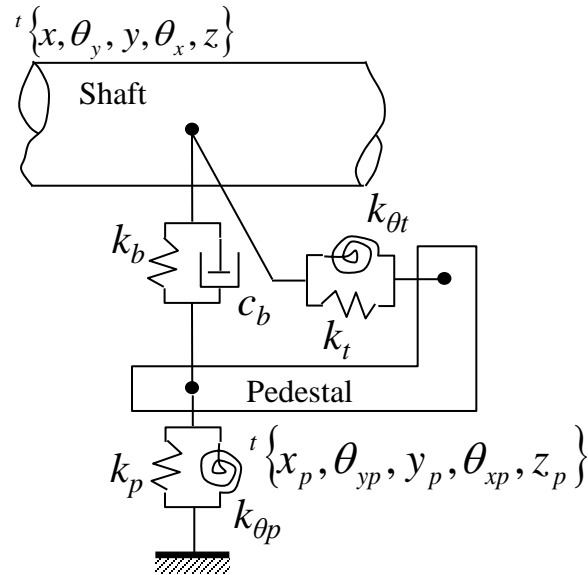
**Unstable
Coupled Vibration**

Image for Coupled Vibration of Lateral and Axial Directions



Model for Coupled Vibration of Lateral and Axial Directions

Axial direction freedom is considered in the rotor and pedestal support system.



k_b : radial bearing stiffness

c_b : radial bearing damping

$k_{\theta t}$: Thrust bearing angular stiffness

k_{zt} : Thrust bearing stiffness

k_p : pedestal stiffness

$k_{\theta p}$: pedestal angular stiffness

-Matrix of shaft stiffness

$$\begin{Bmatrix} F \\ M \end{Bmatrix} = \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} \begin{Bmatrix} x \\ \theta \end{Bmatrix} = K_r \begin{Bmatrix} x \\ \theta \end{Bmatrix}$$

-Matrix of radial bearing stiffness

$$\begin{Bmatrix} F_x \\ F_y \end{Bmatrix} = \begin{bmatrix} k_{xx} & \\ & k_{yy} \end{bmatrix} \begin{Bmatrix} x \\ y \end{Bmatrix} = K_b x$$

-Matrix of thrust bearing stiffness

$$\begin{Bmatrix} M_y \\ M_x \\ F_z \end{Bmatrix} = \begin{bmatrix} k_{\theta xx} & k_{\theta xy} & \\ k_{\theta yx} & k_{\theta yy} & \\ & & k_{tz} \end{bmatrix} \begin{Bmatrix} \theta_y \\ \theta_x \\ z \end{Bmatrix} = K_t x$$

-As a system characteristic matrix

$$\begin{Bmatrix} F_{xp} \\ M_{yp} \\ F_{yp} \\ M_{xp} \\ F_{zp} \end{Bmatrix} = \begin{bmatrix} k_{pxx} & & & & \\ & k_{\theta pyy} & & & \\ & & k_{pyy} & & \\ & & & k_{\theta pxx} & k_{\theta pxz} \\ & k_{\theta pzy} & & k_{\theta pzx} & k_{pz} \end{bmatrix} \begin{Bmatrix} x_p \\ \theta_{yp} \\ y_p \\ \theta_{xp} \\ z_p \end{Bmatrix}$$

Model for Coupled Vibration of Lateral and Axial Directions

The eigenvalue problem is solved.

$$\begin{bmatrix} m & & & & & & & & & & \\ & m & & & & & & & & & \\ & & m_p & & & & & & & & \\ & & & m_p & & & & & & & \\ & & & & I_d & & & & & & \\ & & & & & I_d & & & & & \\ & & & & & & I_{dp} & & & & \\ & & & & & & & I_{dp} & & & \\ & & & & & & & & m & & \\ & & & & & & & & & m_p & \end{bmatrix}
 \begin{Bmatrix} \ddot{x} \\ \ddot{y} \\ \ddot{x}_p \\ \ddot{y}_p \\ \ddot{\theta}_y \\ \ddot{\theta}_x \\ \ddot{\theta}_{yp} \\ \ddot{\theta}_{xp} \\ \ddot{z} \\ \ddot{z}_p \end{Bmatrix}
 +
 \begin{bmatrix} +k_{bxx} & & & & & & & & & & \\ & -k_{bxx} & & & & & & & & & \\ & & +k_{byy} & & & & & & & & \\ & & & -k_{byy} & & & & & & & \\ -k_{bxx} & & & & k_{bxx} & & & & & & \\ & & & & & k_{byy} & & & & & \\ & & & & & & +k_{\theta_{tyy}} & k_{\theta_{tyx}} & -k_{\theta_{tyy}} & -k_{\theta_{txx}} & \\ & & & & & & k_{\theta_{txy}} & +k_{\theta_{ttx}} & -k_{\theta_{txy}} & -k_{\theta_{ttx}} & \\ & & & & & & -k_{\theta_{tyy}} & -k_{\theta_{tyx}} & k_{\theta_{tyy}} & k_{\theta_{txx}} & \\ & & & & & & -k_{\theta_{txy}} & -k_{\theta_{ttx}} & k_{\theta_{txy}} & k_{\theta_{ttx}} & \\ & & & & & & & & & & k_z & -k_z \\ & & & & & & & & & & & k_{zyp} & k_{zxp} & -k_z & k_{zp} \end{bmatrix}
 \begin{Bmatrix} x \\ y \\ x_p \\ y_p \\ \theta_y \\ \theta_x \\ \theta_{yp} \\ \theta_{xp} \\ z \\ z_p \end{Bmatrix} = 0$$

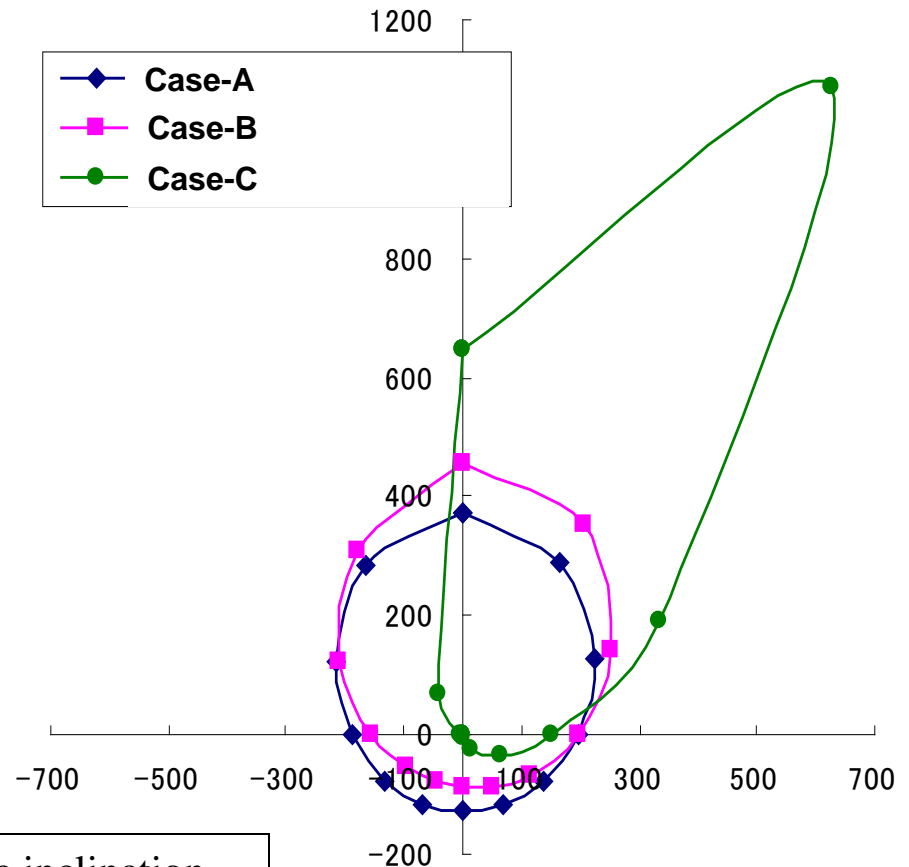
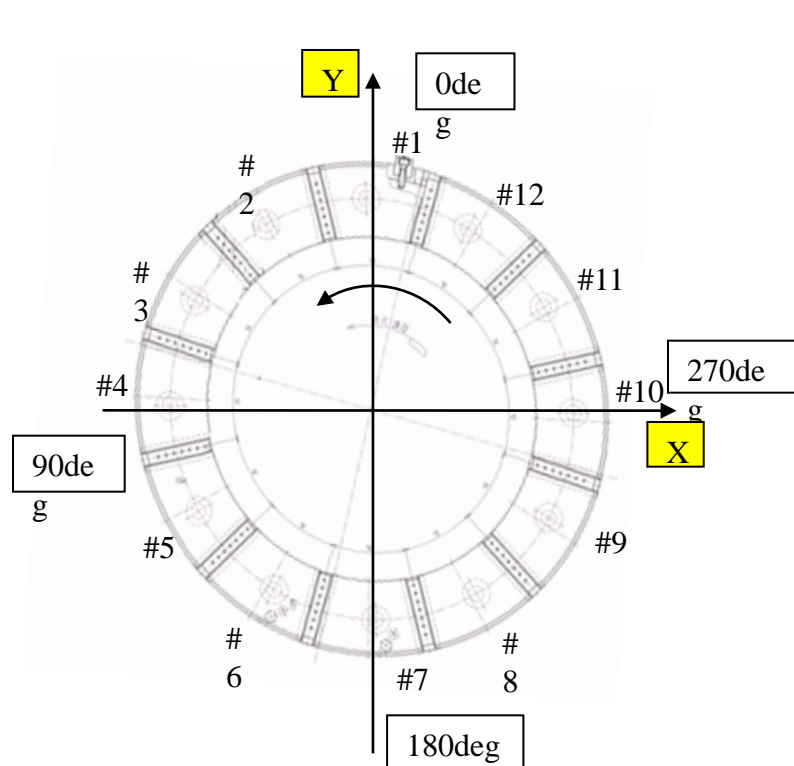
By simply expressed,

$$\begin{bmatrix} m & & & & \\ & I & & & \\ & & m & & \\ & & & m_p & \end{bmatrix}
 \begin{Bmatrix} \ddot{X} \\ \ddot{\Theta} \\ \ddot{z} \\ \ddot{z}_p \end{Bmatrix}
 +
 \begin{bmatrix} K_b & & & \\ & K_{th} & & \\ & & K_{\theta M} & \\ & & & K_{\theta M} \end{bmatrix}
 \begin{Bmatrix} X \\ \Theta \\ z \\ z_p \end{Bmatrix} = 0$$

Symmetry

Only K_{th} is related to the stability because other terms are symmetry.

Simulation Result for Thrust Load Distribution of Pads



Load Distribution of Pads

Case-A: Pads gnaethonically move to the thrust disc inclination.
Case-B: Pads inclination is fixed at the stationary state.
Case-C: Pads are fixed as a flat face.

The cross coupling stiffness becomes larger if the pad static equilibrium are broken up. It would become the driving force of the radial/ axial coupled vibration.

Summary

During shop mechanical running test of steam turbine, radial sub-synchronous vibration was observed. And, at the same sub-synchronous frequency, the axial vibration was also observed.

It was confirmed that the sub-synchronous frequency agrees with the axial natural frequency of the rotor.

Rotor support system was modeled to investigate this interaction of axial and radial direction vibration by considering the axial direction freedom.

As a result of the simulation, the cross coupling stiffness becomes larger if the pad static equilibrium are broken up.

It was confirmed that it becomes the driving force of the radial/ axial coupled vibration.

It is expected that the coupled sub-synchronous vibration of lateral and axial directions can be predicted and evaluated by the simulation.